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Robust control for vibration control systems with dead-zone band and time delay under severe disturbance using adaptive fuzzy neural network

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Abstract

This study proposes a novel adaptive control method to deal with the dead-zone and time delay issues in actuators of vibration control systems. The controller is formulated based on a type-2 fuzzy neural network integrating with a new modification of Riccati-like equation. The developed new type Riccati-like equation is significant as it reduces energy consumption of control inputs to minimum. Two approaches are suggested to improve performance of the system using the basic elements of Riccati equation. In addition, a fuzzy neural network is applied to approximate the unmodeled dynamics and a sliding mode controller is developed to enhance the robustness of the system against uncertainties and disturbances. After proving the stability of the proposed controller via Lyapunov criterion, the effectiveness of the proposed approach is validated based on computer simulation for vibration control of a vehicle seat suspension. It is demonstrated that the unwanted vibrations due to external excitations are well controlled despite of the presence of dead-zone and time delay in actuators. Furthermore, when comparing with other two state-of-the-art robust controllers [23, 36], the proposed controller provides better vibration suppression capacity and requires less energy consumption.

Keywords: Type 2-fuzzy neural network, adaptive control, neural network, dead-zone, time delay, MR damper, vibration control
1. Introduction

Developing new control methodologies for modern devices attracts a lot of attentions from both academia and industry because it is important yet challenging. In general, the control approaches for modern devices can be classified into classical control and modern control. The goal of development of new control mechanism is to satisfy multiple requirements of consumer such as comfortable, human ergonomics, vibration and disturbance suppression, high precision, and so forth. Many great efforts have been spent on designing new control schemes to achieve that goals. For instance, an adaptive control using a super-twisting integrated with adecoupled-terminal sliding mode control was presented in [1], where the sliding surface was designed based on a terminal sliding surface and a super-twisting controller was employed to eliminate chattering. An integral sliding mode control with prescribed performance was developed in [2]. In this approach, a tanh function was used to obtain smooth control efforts and a prescribed function was utilized to ensure the tracking errors are limited within the desirable boundaries. In the design of control system, besides considering disturbances, the dead-zone in actuators and unmodeled dynamics are also significant that need to be taken into account. A methodology to deal with the dead-zone input and unmodeled dynamics was recently studied in [3], where the dead-zone was simply defined as hysteretic properties of the system and the unmodeled dynamics was assumed to be bounded by some constants. Due to its approximation capability, fuzzy logic system (FLS) has been frequently utilized in many practical systems subjected to unmodeled dynamics, parameter variations and non-predictable nonlinear parameters, i.e., actuators and sensor faults. For example, a back-stepping design-based fuzzy approximation has been developed to deal with actuator failures and unknown system dynamics [4]. Moreover, in order to enhance system robustness against external disturbances and input constraints, an adaptive controller was integrated with an $H$-infinity technique [5], in which the constraint inputs are encoded into $H$-infinity performance. Recently, various types of robust controllers have been proposed by combining two or more different controllers to increase the system robustness against the diverse uncertainties and unmodeled dynamics. In general, the stability of these composite or hybrid controllers can be proven based on Lyapunov criterion [6]. A hybrid controller based on PI control, sliding mode control and radial basic function neural
network was proposed in [7]. A problem associated with gain adaptation of conventional sliding mode control was described in [8]. An integration between nonsingular fast terminal sliding model controller and PID controller was also applied to design adaptive controller in [9]. A new two-step distributed adaptive leader-following consensus control of connected Lagrangian systems was described in [10]. A continuous sliding mode control law coupled with an adaptive sliding mode observer was presented in [11]. The nonsingular terminal sliding surface used in [11] was similar to that of the one used in [9]. A study of disturbance rejection based on sliding mode control with prescribed performance for robot manipulators was carried out in [12]. In [13], a PID controller was combined with an $H$-infinity controller based on Riccati-like equation. It is notice that the Riccati-like equation plays an important role in connecting a controller with other controllers without loss of the stability. Therefore, it is often employed for deriving hybrid controllers, which comprise more than two different controllers. Recently, the a static output feedback control for compensating faults of switched system was presented in [14].

In this model [14], the fuzzy control is applied for finding properties of the switched system, and conditions for faults were designed based on the switched nonlinear systems is converted into a switched fuzzy system, and the stability of the system is solved using a linear matrix inequality method. A similar method [14] was also studied in [15] taking the but for missing measurements problem of the system states of the system into consideration. Fault tolerant control under state/noise estimation was studied in [16] using the linear matrix inequality approach was also used for setup conditions [16]. In [17], the a fault tolerant control method [17] was proposed by an adaptation law with matrices relating to ed Hurwitz definition.

Most of actuators in practical applications possess a dead-zone band. Violations of the control inputs with the dead-zone will degrade the system performance significantly. Moreover, time delay of actuators provides adverse effects on system performance. These effects become even much worst for complex or composite control systems, which integrate different controller types and feedback loops. In order to reduce the adverse effects due to the dead-zone band, many approaches have been developed for different types of applications [18-23]. For example, an adaptive fuzzy controller integrated with an $H$-infinity technique was utilized to solve the unknown dead-zone output in [18]. Neural networks approach for the dead-zone issue was studied
in [19]. The dead-zone model was considered as an additional disturbance component and an adaptive robust control was designed to eliminate the effects of dead-zone in [20, 21]. In [22], a modification of backstepping control technique with prescribed performance was proposed to deal with the dead-zone input for air-breathing hypersonic vehicles. An adaptive fuzzy control combined with a modification of Riccati-like equation was studied in [23]. However, this approach has not considered the dead-zone effects in the design.

From the above literature survey, the fuzzy logic system is usually selected as an effective tool for controlling the system subjected to several uncertainties including unmodeled dynamics and parameter variations. This choice is reasonable because the fuzzy logic system can act like a filter in the design of robust control mechanism. Besides the classical fuzzy model, namely type 1 model, the type 2 model has also been employed in the design of control systems [24, 25, 27-30]. Type-2 fuzzy sets and systems generalize standard Type-1 fuzzy sets and systems so that more uncertainty can be handled. Because large data is used to establish the fuzzy rules, clustering method is normally used as the first step to cluster dataset. The FLS and neural network have their own advantages and disadvantages in terms of nonlinear function approximation, so combining FLS and neural network will likely bring more merits while reducing their disadvantages for uncertain nonlinear systems.

Motivated by the aforementioned discussions, in this work, a type 2 fuzzy model associated with a neural network is adopted for modeling system dynamics. The modeled dynamics is then integrated with an adaptive sliding mode controller to improve the system robustness against the system uncertainties and disturbances, which includes both dead-zone input and time delay of the actuators. In summary, the major contributions of this paper can be highlighted as follows:

1. A novel adaptive law for dealing with the dead-zone and delayed time delay phenomenon is proposed.

2. A new modification of Riccati-like equation for both dead-zone phenomenon and time delay of actuators is suggested.

3. The main parameter, which decide how to save energy consumption of the control system, is pointed out in the adaptation laws.
Compared with other state-of-the-art methods [23, 36], the proposed approach provides superior performance.

2. Design of Adaptive Fuzzy Control with Actuator Dead-zone and Time Delay

2.1. Interval Type 2 Fuzzy Model

In this study, the interval type 2 fuzzy neural networks (IT2FNN) model [24,25,27-30] is employed. Before applying, the clustering progress [26] is carried out to decrease the volume of data, and then the rule base of IT2FNN is defined. The \( j \)th If-Then rule can be expressed as follows:

\[
R_j^j: \text{If } h_{i1} \text{ and } ... \text{ and } h_{in} \text{ then } g = a_j^j + \sum_{i=1}^{n} a_i h_i
\]

\[ (1) \]

where, \( H_{ij}^j (i=1,...,n; j=1,...,m) \) are fuzzy sets, \( m \) is the number of rules, and \( a_j^j \) are interval sets.

The defuzzified output is determined by

\[
g_j = \frac{g_j + g_{-j}}{2} = \frac{\Xi_j^j \Phi_j^j + \Xi_{-j}^j \Phi_{-j}^j}{2}
\]

\[ (2) \]

In equation (2), \( \Xi_j^j = [w_{1j} w_{2j} w_{3j}...w_{nj}] \) and \( \Xi_j^j = [w_{1j} w_{2j} w_{3j}...w_{nj}] \) are the weighting vectors, which symbolize the relation between the rule layer and type-reduction. The weights firing strength vectors are given by

\[
\Phi_j^j = \left[ \begin{array}{c}
\frac{f_1}{\sum_{i=1}^{n} f_i} & \frac{f_2}{\sum_{i=1}^{n} f_i} & \frac{f_3}{\sum_{i=1}^{n} f_i} & ... & \frac{f_n}{\sum_{i=1}^{n} f_i}
\end{array} \right]^T, \quad \Phi_j^j = \left[ \begin{array}{c}
\frac{-f_1}{\sum_{i=1}^{n} f_i} & \frac{-f_2}{\sum_{i=1}^{n} f_i} & \frac{-f_3}{\sum_{i=1}^{n} f_i} & ... & \frac{-f_n}{\sum_{i=1}^{n} f_i}
\end{array} \right]^T.
\]

2.2. Adaptive Fuzzy Control with Actuator Dead-zone and Time Delay

In order to formulate a new composite controller, a single-input single-output (SISO) nonlinear system governed by the following equation is considered:

\[
\dot{x} = f(x) + g(x)\phi_k(t) + \Lambda^T x_c + d(t)
\]

\[ (3) \]

where, \( f(x) \in \mathbb{R}^n \) and \( g(x) \in \mathbb{R}^n \) are two unknown non-linear function vectors, \( \phi_k(t) \in \mathbb{R}^l \) is control function, \( d(t) \in \mathbb{R}^n \) is an external disturbance vector,
\( \mathbf{x} = [x_1, x_2, \ldots, x_n]^T = [x_1, \dot{x}_1, \ldots, x_1^{(n-1)}]^T \in \mathbb{R}^n \) is the state vector of the system,

\( \mathbf{x}_r = [x_1^r(t - \tau_1), x_2^r(t - \tau_2), \ldots, x_n^r(t - \tau_n)]^T \) is the state with time delay,

\[
\mathbf{\Lambda} = \begin{pmatrix}
\mathbf{\Lambda}_1(t) & 0 & \cdots & 0 \\
0 & \ddots & \cdots & 0 \\
0 & \cdots & 0 & \mathbf{\Lambda}_n(t)
\end{pmatrix}, \quad \mathbf{\Lambda}_m(t) = [z_m(t) \ 0]^T
\]
is the nonlinear function, which relates time-varying of the system with an assumed boundary \( |z_m(t)| \leq \delta z_m \), where \( \delta z_m \) is an unknown positive constant.

An equivalent form of the system (3) can be expressed as follows:

\[
\dot{\mathbf{x}} = \mathbf{A} \mathbf{x} + \mathbf{B} \left[ \mathbf{f}(\mathbf{x}) + \mathbf{g}(\mathbf{x}) \phi_u(t) + \mathbf{\Lambda}^T \mathbf{x}_r + \mathbf{d}(t) \right]
\]  

(4)

where, \( \mathbf{A} = \begin{bmatrix} 0 & 1 & \cdots & 0 \\ 0 & 0 & \cdots & 0 \\ \vdots & \vdots & \cdots & \vdots \\ 0 & 0 & \cdots & 1 \end{bmatrix} \), \( \mathbf{B} = \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 1 \end{bmatrix} \)

The control function \( \phi_u(t) \) in (4) can be designed as follows:

\[
\phi_u(t) = \beta u + \varphi(u) + \delta(\dot{u})
\]  

(5)

where, \( \beta \) is the parameter related to the slope of the dead-zone performance, as shown in Figure 1, \( u \) is the control input, and the functions \( \varphi(u) \) and \( \delta(\dot{u}) \) are defined as follows [19]:

\[
\varphi(u) = \begin{cases}
-\beta u & \text{for } u \geq \theta^+ \\
-\beta \theta^+ & \text{for } \theta^- \leq u \leq \theta^+ \\
-\beta \theta^- & \text{for } u \leq \theta^- 
\end{cases}
\]  

(6)

\[
\delta(\dot{u}) = \beta |\dot{u}| \text{sign}(\dot{u})
\]  

(7)

where, \( \beta \) is a chosen constant gain. The values of Eqs. (6) and (7) are assumed to be bounded by \( |\varphi(u)| \leq \Delta \varphi \) and \( |\delta(\dot{u})| \leq \Delta \delta \), respectively. The tracking reference function is determined as:

\[
\dot{\mathbf{x}}_d = \mathbf{A} \mathbf{x}_d + \mathbf{B} \dot{\mathbf{x}}_d
\]  

(8)

where, \( \mathbf{x}_d = [x_{d1}, x_{d2}, \ldots, x_{dn}]^T = [x_{d1}, \dot{x}_{d1}, \ldots, x_{d1}^{(n-1)}]^T \in \mathbb{R}^n \). The error dynamic equation is defined as \( \mathbf{e} = \mathbf{x} - \mathbf{x}_d \). The derivative of this error can be calculated as:

\[
\dot{\mathbf{e}} = \dot{\mathbf{x}} - \dot{\mathbf{x}}_d = \mathbf{A} \mathbf{x} + \mathbf{B} \left[ \mathbf{f}(\mathbf{x}) + \mathbf{g}(\mathbf{x}) \phi_u(t) + \mathbf{\Lambda}^T \mathbf{x}_r + \mathbf{d}(t) \right] - (\mathbf{A} \mathbf{x}_d + \mathbf{B} \dot{\mathbf{x}}_d)
\]
\[
\phi(u) = \theta\beta - \phi' \delta + \phi(u) + \delta(u)
\]

Figure 1. Nonlinear dead-zone band model.

From Eq. (9), a new feedback gain matrix \( K = \text{diag} \{ K_1, \ldots, K_n \} \) with \( K_i = [K_{ii} \ K_{i2}]^T \) is a chosen vector such that all roots of the characteristic polynomial of \( A_i = A - BK_i^T \), \( i = 1, \ldots, n \) are in the open left-half plane. From this definition, a new matrix \( \Lambda = \text{diag} \{ \Lambda_1, \ldots, \Lambda_n \} \) is declared. Eq. (9) is then rewritten as follows:

\[
\dot{e} = A\dot{e} - BK^T e + B[K^T e + f(x) + g(x)[\beta u + \varphi(u) + \delta(u)] + \Lambda^T x_e + d(t) - \tilde{x}_d]
\]

Using the result of fuzzy neural networks model (2), the nonlinear functions \( f(x) \) and \( g(x) \) can be expressed as:

\[
\hat{f}(x) = \Xi_f(x) \Phi_f, \quad \hat{g}(x) = \Xi_g(x) \Phi_g
\]

When the fuzzy neural networks model is used, the minimum approximation errors of \( f(x) \) and \( g(x) \) are defined as follows:

\[
\Delta_f(x) = f(x) - \hat{f}(x | \Phi^*_f), \quad \Delta_g(x) = g(x) - \hat{g}(x | \Phi^*_g)
\]

where, \( \Delta_f(x) \) and \( \Delta_g(x) \) are the minimum approximation errors. These errors will be compensated by using adaptation laws. The vectors \( \Phi^*_f \) and \( \Phi^*_g \) are defined as follows:

\[
\Phi^*_f = \arg \min_{\Phi_f \in \Omega_f} \left\{ \sup_{x \in \Lambda^T_f} |f(x) - \hat{f}(x | \Phi_f)| \right\}, \quad \Phi^*_g = \arg \min_{\Phi_g \in \Omega_g} \left\{ \sup_{x \in \Lambda^T_g} |g(x) - \hat{g}(x | \Phi_g)| \right\}
\]

To support the next analysis steps, two new definitions are introduced as follows:

\[
\Phi_f = \Phi^*_f - \Phi_f, \quad \Phi_g = \Phi^*_g - \Phi_g
\]

Using the results in Eqs. (11) and (14), Eq. (10) is rewritten as follows:
\[
\dot{e} = A_\Delta e + B \left[ K^T e + f(x) - \hat{f}(x) + \left( g(x) - \hat{g}(x) + \tilde{g}(x) \right) \left[ \beta u + \phi(u) + \delta(u) \right] \right] + \\
+ \Lambda^T x_r + d(t) - \bar{x}_d.
\]

\[
\Rightarrow \dot{e} = A_\Delta e + B \left[ K^T e + \mathcal{Z}(x) \Phi_f + \hat{f}(x) + \left( \mathcal{Z}(x) \Phi_g + \hat{g}(x) \right) \left[ \beta u + \phi(u) + \delta(u) \right] \right] + \\
+ \Lambda^T x_r + d(t) - \bar{x}_d.
\]

\[
\Rightarrow \dot{e} = A_\Delta e + B \left[ K^T e + \mathcal{Z}(x) \Phi_f + \hat{f}(x) + \mathcal{Z}(x) \Phi_g + \left[ \beta u + \phi(u) + \delta(u) \right] \right] + \\
+ \Lambda^T x_r + d(t) - \bar{x}_d + D
\]

(15)

where, \( D = \Delta f + \Delta g \left[ \beta u + \phi(u) + \delta(u) \right] \) is the minimum approximation error associated to dead-zone phenomenon. The proposed controllers are now described as below.

**Proposed Control 1.**

In this proposed control 1, the state with time delay \( x_r = [x_1'(t-r_1) \ x_2'(t-r_2) \ \ldots \ x_n'(t-r_n)]^T \) is neglected. The main control \( u \) is determined as follows:

\[
u = u_1 + u_2 + u_3 = \sum_{i=1}^{3} u_i
\]

(16)

where, \( u_i \) is the equivalent control, \( u_2 \) is the first robustness control related to the Riccati-like equation, and \( u_3 \) is the second robustness control related to the dead-zone band and the minimum approximation error \( D \) of the system. These control inputs are designed as follows:

\[
u_1 = -\frac{1}{\beta^2 \tilde{g}(x)} \left( f(x) - \bar{x}_d + K^T e \right)
\]

(17)

\[
u_2 = -\frac{1}{\beta^2 \tilde{g}(x)} R^{-1} B^T Pe
\]

(18)

\[
u_3 = -\frac{1}{\beta_{\text{min}}^\phi} (\Delta \phi + \Delta \delta) \text{sign} (\tilde{g}(x)) - \frac{1}{\beta_{\text{min}}^\phi \tilde{g}(x)} D
\]

(19)

where, \( \Delta \phi \) is the upper bound value of \( \|\phi(u)\| \), \( \Delta \delta \) is the upper bound value of \( \|\delta(u)\| \), \( \beta_{\text{min}}^\phi \) is the parameter related to the dead-zone model. The balance Riccati-like equation is given by

\[
PA + A^T P + Q + PB \left( \frac{2}{\rho^2} I - \beta_{\text{min}}^\phi R^{-1} \right) B^T P = 0
\]

(20)

where, \( 0 < \rho < 1 \), \( I \) is the identity matrix \((n \times n)\), \( Q = Q^T \). \( R \) denotes the robust \( H \)-infinity gain.

**Theorem 1.** The stability and robustness of the proposed controller for the dead-zone system are guaranteed if the control function \( u \) is design as in (16) with its elements are implemented as in Eqs.(17), (18), (19), and the adaptation laws are given below.
\[ \dot{\Phi}_f = -\gamma_f \Xi_f^T B^T Pe \] (22)
\[ \dot{\Phi}_g = -\gamma_g \Xi_g^T B^T Pe(u + \Delta \varphi + \Delta \delta) \] (23)
\[ \dot{D}_\beta = -\frac{\gamma_\beta}{\beta} B^T Pe \] (24)
\[ \dot{\beta}_\beta = -\gamma_\beta \left[ e^T PB \left( \hat{f}(x) - \hat{x}_d + K^T e \right) \right] \] (25)

where, \( \gamma_f, \gamma_g, \gamma_D, \gamma_\beta \) are chosen constants. It is noted that \( \hat{\beta}_\beta = \beta_\beta - \dot{\beta}_\beta, \beta_\beta = \frac{1}{\beta}, \dot{\beta}_\beta = \frac{1}{\beta^2} \).

\[ \hat{D}_\beta = \frac{1}{\beta} D \rightarrow \bar{D}_\beta = D - \hat{D}_\beta; \quad \gamma_f, \gamma_g, \gamma_\beta, \gamma_D \] are chosen constants.

**Proof:** To prove the theorem, a Lyapunov function candidate is proposed as follows:

\[ L_v = \frac{1}{2\beta} e^T Pe + \frac{1}{2\beta \gamma_f} \Phi_f^T \Phi_f + \frac{1}{2\beta \gamma_g} \Phi_g^T \Phi_g + \frac{1}{2\gamma_\beta} \beta_\beta^2 + \frac{1}{2\gamma_D} \dot{D}_\beta^T \dot{D}_\beta \] (26)

where \( \beta_\beta = \beta_\beta - \dot{\beta}_\beta, \beta_\beta = \frac{1}{\beta}, \dot{\beta}_\beta = \frac{1}{\beta^2} \).

The derivative of Eq. (26) is obtained as:

\[ \dot{L}_v = \frac{1}{2\beta} e^T Pe + \frac{1}{2\beta \gamma_f} \Phi_f^T \Phi_f + \frac{1}{2\beta \gamma_g} \Phi_g^T \Phi_g + \frac{1}{\gamma_\beta} \dot{\beta}_\beta \beta_\beta + \frac{1}{\gamma_D} \dot{D}_\beta^T \dot{D}_\beta \] (27)

Using the results in Eq.(20), Eq.(27) is rewritten as follows:

\[ \dot{L}_v = \frac{1}{2\beta} e^T \left( A^T P + PA \right) e + \frac{1}{\beta} e^T PB \Xi_f^T \Phi_f + \frac{1}{\beta} e^T PB \Xi_g^T \Phi_g + \frac{1}{\beta} e^T PB \Xi_D^T \dot{D}_\beta \] (28)

Using the adaptation laws (21) and (22), Eq.(28) is rewritten as follows:

\[ \dot{L}_v \leq \frac{1}{2\beta} e^T \left( A^T P + PA \right) e + \frac{1}{\beta} e^T PB \left( \hat{f}(x) - \hat{x}_d + K^T e + \beta_\beta \hat{g}(x)(u_1 + u_2 + u_3) \right) + \frac{1}{\beta} e^T PB \dot{D}_\beta \] (29)

Using the results in Eq.(20), Eq.(29) is rewritten as follows:
\[
\dot{L}_v = \frac{1}{2\beta} e^T \left( -Q - \frac{2}{\rho^2} PBB^T P + \beta_{\text{min}} PBR^{-1} B^T P \right) e + \frac{1}{\beta} e^T P B \left( \hat{f}(x) - \hat{x}_d + K^T e + \beta \hat{g}(x)(u_1 + u_2 + u_3) + \hat{g}(x) (\Delta \phi + \Delta \delta) \right) \\
+ \frac{1}{\beta} e^T P B D + \frac{1}{\gamma_\beta} \hat{\beta}_\mu \hat{\beta}_\mu + \frac{1}{\gamma_D} \hat{D}_\mu \hat{D}_\mu \\
\Rightarrow \dot{L}_v \leq \frac{1}{2\beta} e^T \left( -Q - \frac{2}{\rho^2} PBB^T P + \beta_{\text{min}} PBR^{-1} B^T P \right) e + \frac{1}{\beta} e^T P B \left( \hat{f}(x) - \hat{x}_d + K^T e + \beta \hat{g}(x)(u_1 + u_2 + u_3) + \hat{g}(x) (\Delta \phi + \Delta \delta) \right) \\
+ e^T P B \hat{g}(x) (u_1 + u_2 + u_3) + \| e^T P B \| \| g(x) \| \Delta \phi + \Delta \delta + \frac{1}{\beta} e^T P B D + \frac{1}{\gamma_\beta} \hat{\beta}_\mu \hat{\beta}_\mu + \frac{1}{\gamma_D} \hat{D}_\mu \hat{D}_\mu \\
\Rightarrow \dot{L}_v \leq \frac{1}{2\beta} e^T \left( -Q - \frac{1}{\rho^2} PBB^T P + \beta_{\text{min}} PBR^{-1} B^T P \right) e + \frac{1}{\beta} e^T P B \left( \hat{f}(x) - \hat{x}_d + K^T e + \beta \hat{g}(x)(u_1 + u_2 + u_3) + \hat{g}(x) (\Delta \phi + \Delta \delta) \right) \\
+ e^T P B \hat{g}(x) (u_1 + u_2 + u_3) + \| e^T P B \| \| g(x) \| \Delta \phi + \Delta \delta + \frac{1}{\beta} e^T P B D + \frac{1}{\gamma_\beta} \hat{\beta}_\mu \hat{\beta}_\mu + \frac{1}{\gamma_D} \hat{D}_\mu \hat{D}_\mu \\
\Rightarrow \dot{L}_v \leq \frac{1}{2\beta} e^T \left( -Q - \frac{1}{\rho^2} PBB^T P + \beta_{\text{min}} PBR^{-1} B^T P \right) e + \frac{1}{\beta} e^T P B \left( \hat{f}(x) - \hat{x}_d + K^T e + \beta \hat{g}(x)(u_1 + u_2 + u_3) + \hat{g}(x) (\Delta \phi + \Delta \delta) \right) \\
+ e^T P B \hat{g}(x) (u_1 + u_2 + u_3) + \| e^T P B \| \| g(x) \| \Delta \phi + \Delta \delta + \frac{1}{\beta} e^T P B D + \frac{1}{\gamma_\beta} \hat{\beta}_\mu \hat{\beta}_\mu + \frac{1}{\gamma_D} \hat{D}_\mu \hat{D}_\mu \\
= \frac{1}{2\beta} e^T \left( -Q - \frac{1}{\rho^2} PBB^T P + \beta_{\text{min}} PBR^{-1} B^T P \right) e + 
\]
\begin{align*}
+ \beta_\rho e^T \text{PB} \left( \dot{f}(x) - \ddot{x}_d + K^T e \right) - \hat{\beta}_\rho e^T \text{PB} \left( \hat{f}(x) - \ddot{x}_d + K^T e \right) + \hat{\beta}_\rho e^T \text{PB} \left( \dot{f}(x) - \ddot{x}_d + K^T e \right) - \\
- e^T \text{PB} \hat{g}(x) \left( \frac{\beta_\rho}{\hat{g}(x)} \left( \dot{f}(x) - \ddot{x}_d + K^T e \right) \right) - e^T \text{PB} \hat{g}(x) \left( \frac{\beta_\rho}{\hat{g}(x)} \hat{R} B' P e \right) + \\
- e^T \text{PB} \hat{g}(x) \left( \frac{1}{\beta_{\min}} (\Delta \phi + \Delta \delta) \text{sign} \left( \hat{g}(x) \right) \right) - e^T \text{PB} \hat{g}(x) \left( \frac{1}{\beta_{\min}} \hat{D}_\rho \hat{D}_\rho \right) + \\
\frac{\Delta \phi + \Delta \delta}{\beta_{\min}} \text{sign} \left( \hat{g}(x) \right) B' P e + \frac{1}{\beta} e^T \text{PBD} + \frac{1}{\gamma_{\beta}} \hat{\beta}_\rho \hat{\beta}_\rho + \frac{1}{\gamma_{D}} \hat{D}_\rho \hat{D}_\rho \quad (31)
\end{align*}

Eq.(31) is rewritten as follows:

\begin{align*}
\dot{L}_e & \leq \frac{1}{2\beta} e^T \left( -Q + \frac{1}{\rho^2} \text{PBB}' P + \beta_{\min} \text{PBR}^{-1} \text{B}' P \right) e + \\
+ \left( \beta_\rho - \hat{\beta}_\rho \right) e^T \text{PB} \left( \dot{f}(x) - \ddot{x}_d + K^T e \right) + \hat{\beta}_\rho e^T \text{PB} \left( \dot{f}(x) - \ddot{x}_d + K^T e \right) - \hat{\beta}_\rho e^T \text{PB} \left( \dot{f}(x) - \ddot{x}_d + K^T e \right) - \\
- \hat{\beta}_\rho e^T \text{PBR}^{-1} \text{B}' P e - \frac{1}{\beta_{\min}} (\Delta \phi + \Delta \delta) \text{sign} \left( \hat{g}(x) \right) B' P e - \frac{1}{\beta} e^T \text{PBD} + \frac{1}{\gamma_{\beta}} \hat{\beta}_\rho \hat{\beta}_\rho + \frac{1}{\gamma_{D}} \hat{D}_\rho \hat{D}_\rho 
\end{align*}

Eq. (32) is rewritten as follows:

\begin{align*}
L_e & \leq \frac{1}{2\beta} e^T \left( -Q + \frac{1}{\rho^2} \text{PBB}' P \right) e + \hat{\beta}_\rho e^T \text{PBR}^{-1} \text{B}' P e + \hat{\beta}_\rho e^T \text{PB} \left( \dot{f}(x) - \ddot{x}_d + K^T e \right) - \\
- \hat{\beta}_\rho e^T \text{PBR}^{-1} \text{B}' P e - \frac{1}{\beta_{\min}} (\Delta \phi + \Delta \delta) \text{sign} \left( \hat{g}(x) \right) B' P e + \frac{\Delta \phi + \Delta \delta}{\beta_{\min}} \text{sign} \left( \hat{g}(x) \right) B' P e + \\
+ \frac{1}{\beta} e^T \text{PB} \left( D - \hat{D}_\rho \right) + \frac{1}{\gamma_{\beta}} \hat{\beta}_\rho \hat{\beta}_\rho + \frac{1}{\gamma_{D}} \hat{D}_\rho \hat{D}_\rho 
\end{align*}

Using the adaptation laws (23-24), Eq.(33) is rewritten as follows:

\begin{align*}
\dot{L}_e & \leq - \frac{1}{2\beta} e^T Q e + \frac{1}{2\beta^2} e^T \text{PBB}' P e = - \frac{1}{2} \beta_\rho e^T Q e + \frac{1}{2\beta^2} \beta_\rho e^T \text{PBB}' P e \\
\Rightarrow \dot{L}_e & \leq - \frac{1}{2} \beta_\rho |e^T Q e| + \frac{1}{2} \beta_\rho |o^2|
\end{align*}
where \( \frac{1}{\rho^2} e^T PBB^T P e \leq \omega^2 \) with \( \omega^2 \) is the upper bound value of \( \frac{1}{\rho^2} e^T PBB^T P e \). The above equation cannot be used to conclude the system stability. Hence, it needs to be integrated from \( t = 0 \) to \( t = T \):

\[
L_v(0) - L_v(T) + \frac{1}{2} \left| \beta \right| \left[ \int_0^T w^2 \, dt \right] \geq \frac{1}{2} \left| \beta \right| \left[ \int_0^T \rho \, e^T Q \rho \, dt \right]
\]

(34)

where, \( L_v(0) = \frac{1}{2\beta} \left[ \rho^T (0) Pe(0) + \frac{1}{2\gamma_f} \Phi_f^T (0) \Phi_f (0) + \frac{1}{2\gamma_g} \Phi_g^T (0) \Phi_g (0) + \frac{1}{2\gamma_D} \tilde{D}_D^T (0) \tilde{D}_D (0) + \frac{1}{\gamma_v} \hat{v}^T (0) \hat{v} (0) \right] \). The value \( L_v(T) \) is always positive, so Eq. (34) is determined as:

\[
L_v(0) + \frac{1}{2} \left| \beta \right| \left[ \int_0^T w^2 \, dt \right] \geq \frac{1}{2} \left| \beta \right| \left[ \int_0^T \rho \, e^T Q \rho \, dt \right] \geq 0
\]

(35)

The proof is completed. Figure 2 illustrates the block diagram of the proposed control 1. Firstly, the main controls and the adaptation laws will use outputs of the fuzzy neural networks model, the modified Riccati-like equation, and the dead-zone band for calculation. The projection algorithm [38] is also applied for the adaptation laws to improve the process. The output of the main control is the input of the plant and the observer. The dynamic states of the plant after controlling will be compared with the state variables of the observer, and the results of this comparison are the basis for the performance evaluation of the proposed control. The error of this comparison is used as input to updated input of the fuzzy neural networks model and the adaptation laws. In addition, the states of the plant are also updated into the fuzzy model. The process is continuously updated until the system states of the plant obtain stable convergence.

**Remark 1.** The update law \( \tilde{\beta} \beta \) in (25) will be used to replace the constant parameter \( \beta \) in the control inputs \( u_1, u_2, u_3 \) (in Eqs. (17-19)) to ensure the robustness of the proposed controller. This is a new approach to adapt the gain. Compared to the existing adaptation laws, this modification reduces energy consumption of the control input to minimum value.
Figure 2. Block diagram of the proposed algorithm 1.

**Proposed Control 2.**

In the proposed control 2, full phenomena of the system including dead-zone phenomenon and delay time in actuators presented in Eq. (1) are used. Now, the main control $u$ is determined as follows:

$$u = u_1 + u_2 + u_3 + u_4 = \sum_{i=1}^{4} u_i$$  \hspace{1cm} (36)

where, $u_i$ is the equivalent control, $u_2$ is the first robustness control related to the Riccati-like equation, and $u_3$ is the second robustness control input, which is used to deal with the dead-zone band, the Riccati-like equation and the minimum approximation error $D$ of the system, and $u_4$ is the third robustness control input, which is used to tackle the effects of the time delay. These controllers are designed as follows:

$$u_1 = -\frac{1}{\beta_2^2 \hat{g}(x)} \left( \hat{f}(x) - \hat{x}_d + K^T e \right)$$  \hspace{1cm} (37)

$$u_2 = -\frac{1}{\beta_2^2 \hat{g}(x)} R^{-1} B^T P e$$  \hspace{1cm} (38)

$$u_3 = -\frac{1}{\beta_{min}} (\Delta \phi + \Delta \delta) \text{sign}(\hat{g}(x)) - \frac{1}{\beta_2^2 \hat{g}(x)} D$$  \hspace{1cm} (39)
\[ u_4 = - \frac{1}{\beta_{\min} \hat{g}(x)} \hat{B}^T P e \]  

(40)

where, \( \Delta \phi \) is the upper bound value of \( \| \phi(u) \| \), \( \Delta \delta \) is the upper bound value of \( \| \delta(\dot{u}) \| \). The balance Riccati-like equation is given by

\[
PA + A^T P + Q + PB \left( \frac{2}{\rho_1} I - \beta_{\min 1} R^{-1} \right) B^T P + PB \left( \frac{2}{\rho_2} I - \beta_{\min 2} R^{-1} \right) B^T P = 0
\]

(41)

where, \( 0 < \rho_1 < 1, 0 < \rho_2 < 1 \), \( I \) is the identity matrix \((n \times n)\), \( Q = Q^T \), \( R \) denotes the robust H-infinity technique gain, \( \beta_{\min 1} \) is the parameter related to the dead-zone model, \( \beta_{\min 2} \) is the parameter related to the time delay model.

**Theorem 2.** The stability and robustness of the control system with the dead-zone band and time delay are guaranteed if the control input \( u \) is designed as in (36) with its elements shown in Eqs.(37)-(40), and the adaptation laws for these control inputs are chosen as

\[
\dot{\Phi}_f = -\gamma_f \Xi_f B^T P e
\]

(42)

\[
\dot{\Phi}_s = -\gamma_s \Xi_s B^T P e (u + \Delta \phi + \Delta \delta)
\]

(43)

\[
\dot{D}_\beta = -\frac{\gamma_D}{\beta} B^T P e
\]

(44)

\[
\dot{\beta}_\beta = -\gamma_\beta \left[ e^T PB \left( \hat{f}(x) - \ddot{x}_d + K^T e \right) \right]
\]

(45)

\[
\dot{v}^T = -\frac{\gamma_v}{\beta_{\min 2}} \left( B^T P e \right) \left( e^T PB \right)
\]

(46)

where, \( \gamma_f, \gamma_s, \gamma_D, \gamma_\beta, \gamma_v \) are chosen constants, \( \beta \) is the parameter related to the slope of the dead-zone.

**Proof:** To prove the theorem, a Lyapunov function candidate is proposed as follows:

\[
L_v = \frac{1}{2\beta} e^T Pe + \frac{1}{2\beta\gamma_f} \Phi_f^T \Phi_f + \frac{1}{2\beta\gamma_s} \Phi_s^T \Phi_s + \frac{1}{2\gamma_\beta} \beta_{\beta}^2 + \frac{1}{2\gamma_D} \hat{D}_\beta^2 + \frac{1}{2\gamma_v} \hat{v}^T \hat{v}
\]

(47)

where, \( \beta_{\beta} = \beta_\beta - \hat{\beta}_\beta, \beta_\beta = \frac{1}{\beta}, \hat{\beta}_\beta = \frac{1}{\beta^2}, \hat{D}_\beta = \frac{1}{\beta} D, \hat{D}_\beta = D - \hat{D}_\beta, v = v - \hat{v}; \gamma_f, \gamma_s, \gamma_\beta, \gamma_D, \gamma_v \) are chosen constants.

The derivative of Eq. (47) is obtained as:

\[
\dot{L}_v = \frac{1}{2\beta} e^T Pe + \frac{1}{2\beta} \dot{e}^T Pe + \frac{1}{2\beta\gamma_f} \dot{\Phi_f}^T \Phi_f + \frac{1}{2\beta\gamma_s} \dot{\Phi_s}^T \Phi_s + \frac{1}{2\gamma_\beta} \dot{\beta}_{\beta}^2 + \frac{1}{2\gamma_D} \dot{\hat{D}}_\beta^2 + \frac{1}{2\gamma_v} \dot{\hat{v}}^2 \hat{v} \]

(48)

Using Eq.(40), Eq.(48) is rewritten as follows:
\[
\dot{L}_p = \frac{1}{2\beta} e^T (A^T P + PA) e + \frac{1}{\beta} e^T P B \dot{\Phi} + \frac{1}{\beta} e^T P B \dot{\Phi}_\gamma + \frac{1}{\beta} e^T P B \dot{\Phi}_\eta + \frac{1}{\beta} e^T P B \dot{\Phi}_\xi + \frac{1}{\beta} e^T P B \dot{\Phi}_\zeta (\beta u + \Delta \varphi + \Delta \delta) + \\
+ \frac{1}{\beta} e^T P B \ddot{f}(x) + \frac{1}{\beta} e^T P B \ddot{g}(x) (\beta u + \Delta \varphi + \Delta \delta) + \frac{1}{\beta} e^T P B d + \frac{1}{\beta} e^T P B D - \frac{1}{\beta} e^T P B \ddot{x}_d + \\
+ \frac{1}{\beta} e^T P B \Lambda^T x + \frac{1}{\beta} e^T P B \ddot{\Phi} + \frac{1}{\beta} e^T P B \ddot{\Phi}_\gamma + \frac{1}{\beta} e^T P B \ddot{\Phi}_\eta + \frac{1}{\beta} e^T P B \ddot{\Phi}_\xi + \frac{1}{\beta} e^T P B \ddot{\Phi}_\zeta + \frac{1}{\gamma_\nu} \dot{\nu}^T \dot{\nu} \tag{49}
\]

Using the adaptation laws (42) and (43), Eq.(49) is rewritten as follows:

\[
\dot{L}_p \leq \frac{1}{2\beta} e^T (A^T P + PA) e + \frac{1}{\beta} e^T P B \left( \ddot{f}(x) - \ddot{x}_d + K^T e + \beta \ddot{g}(x) (u_1 + u_2 + u_3 + u_4) + \right) + \\
+ \frac{1}{\beta} e^T P B D + \frac{1}{\beta} e^T P B \Lambda^T x + \frac{1}{\gamma_\nu} \dot{\nu}^T \dot{\nu} \tag{50}
\]

Using Eq.(41), Eq.(50) is rewritten as follows:

\[
\dot{L}_p = \frac{1}{2\beta} e^T \left( -Q - \frac{2}{\rho_1} P B B^T P + \beta_{min1} P B R^{-1} B^T P - \right) + \frac{1}{\beta} e^T P B \left( \ddot{f}(x) - \ddot{x}_d + K^T e + \right) + \\
+ \frac{1}{\beta} e^T P B D + \frac{1}{\beta} e^T P B \Lambda^T x + \frac{1}{\gamma_\nu} \dot{\nu}^T \dot{\nu} \tag{51}
\]

Using Eqs.(37)-(40), Eq.(51) is rewritten as follows:

\[
\dot{L}_p \leq \frac{1}{2\beta} e^T \left( -Q - \frac{2}{\rho_1} P B B^T P + \beta_{min1} P B R^{-1} B^T P + \right) e + \frac{1}{\beta} e^T P B \left( \ddot{f}(x) - \ddot{x}_d + K^T e + \right) + \\
+ e^T P B \ddot{g}(x) (u_1 + u_2 + u_3 + u_4) + \|e^T P B \| \frac{\Delta \varphi + \Delta \delta}{\beta} + \\
+ \frac{1}{\beta} e^T P B D + \frac{1}{\beta} e^T P B \Lambda^T x + \frac{1}{\gamma_\nu} \dot{\nu}^T \dot{\nu} \tag{52}
\]
\[ + e^\top \text{PB} \hat{g}(x) \left( - \frac{1}{\beta^2} \hat{g}(x) \left( \hat{f}(x) - \hat{x}_d + K^\top e \right) \right) + e^\top \text{PB} \hat{g}(x) \left( - \frac{1}{\beta^2} \hat{g}(x) R^\top B^\top Pe \right) + \]

\[ + e^\top \text{PB} \hat{g}(x) \left( - \frac{1}{\beta^2} \hat{g}(x) \left( \Delta \phi + \Delta \delta \right) \text{sign} \left( \hat{g}(x) \right) - \frac{1}{\beta^2} \hat{g}(x) D \right) + e^\top \text{PB} \hat{g}(x) \left( - \frac{1}{\beta^2} \hat{g}(x) \right) \nu^\top B^\top Pe + \]

\[ + \| \text{PB} \| \hat{g}(x) \left( \frac{\Delta \phi + \Delta \delta}{\beta} + 1 \right) e^\top \text{PBD} + \frac{1}{\beta} e^\top \text{PBA}^\top x_t + \frac{1}{\gamma} \hat{\beta}_\mu \hat{\beta}_\mu + \frac{1}{\gamma_e} \hat{D}_\mu \hat{D}_\mu + \frac{1}{\gamma_e} \hat{v}^\top \hat{v} \]

\[ \Leftrightarrow \hat{l}_o \leq \frac{1}{2\beta} e^\top \left( -Q + \frac{1}{\rho_1^2} \text{PBB}^\top P + \beta_{\min 1} \text{PBR}^{-1} B^\top P + \right) e + \]

\[ + \frac{1}{\beta} e^\top \text{PB} \hat{f}(x) - \hat{x}_d + K^\top e + \frac{1}{\beta} e^\top \text{PB} \left( \hat{f}(x) - \hat{x}_d + K^\top e \right) + \frac{1}{\beta} e^\top \text{PB} \hat{f}(x) - \hat{x}_d + K^\top e \]

\[ - e^\top \text{PB} \hat{g}(x) \left( \hat{\beta}_\mu \hat{g}(x) \hat{f}(x) - \hat{x}_d + K^\top e \right) - e^\top \text{PB} \hat{g}(x) \left( \frac{1}{\beta^2} \hat{g}(x) \right) + \]

\[ - e^\top \text{PB} \hat{g}(x) \left( \Delta \phi + \Delta \delta \text{sign} \left( \hat{g}(x) \right) - e^\top \text{PB} \hat{g}(x) \left( \frac{1}{\beta^2} \hat{g}(x) \right) \hat{D}_\mu \right) + \frac{1}{\beta_{\min 2}} \hat{g}(x) \nu^\top B^\top Pe + \]

\[ + \frac{1}{\beta_{\min 1}} \Delta \phi + \Delta \delta \text{sign} \left( \hat{g}(x) \right) B^\top Pe + \frac{1}{\beta} e^\top \text{PBD} + \frac{1}{\beta} e^\top \text{PBA}^\top x_t + \frac{1}{\gamma} \hat{\beta}_\mu \hat{\beta}_\mu + \frac{1}{\gamma_e} \hat{D}_\mu \hat{D}_\mu + \frac{1}{\gamma_e} \hat{v}^\top \hat{v} \]

Now, Eq.(52) is rewritten as follows:

\[ \hat{l}_o \leq \frac{1}{2\beta} e^\top \left( -Q + \frac{1}{\rho_1^2} \text{PBB}^\top P + \left( \beta_{\min 1} + \beta_{\min 2} \right) \text{PBR}^{-1} B^\top P \right) e + \]

\[ + \left( \beta - \hat{\beta} \right) e^\top \text{PB} \left( \hat{f}(x) - \hat{x}_d + K^\top e \right) + \hat{\beta}_\mu e^\top \text{PB} \left( \hat{f}(x) - \hat{x}_d + K^\top e \right) - \hat{\beta}_\mu e^\top \text{PB} \left( \hat{f}(x) - \hat{x}_d + K^\top e \right) - \]
\[
\begin{aligned}
-\hat{\beta}_\rho e^T PBR^{-1} B^T Pe - \frac{1}{\beta_{\min 1}} (\Delta \varphi + \Delta \delta) \text{sign}(\hat{g}(x)) B^T Pe - \frac{1}{\beta_{\min 2}} \hat{\beta}_\rho e^T PBD - \frac{1}{\gamma_{\delta}} \hat{\nu} B^T Pe e^T PB + \\
+ \frac{\Delta \varphi + \Delta \delta}{\beta_{\min 1}} \text{sign}(\hat{g}(x)) B^T Pe + \frac{1}{\beta_{\min 2}} e^T PBD + \frac{1}{\beta_{\min 2}} \nu B^T Pe e^T PB + \frac{1}{\gamma_{\rho}} \hat{\beta}_\rho \beta_{\rho} + \frac{1}{\gamma_{D}} \hat{D}_\rho \hat{D}_\rho + \frac{1}{\gamma_{\nu}} \hat{\nu} \hat{\nu}
\end{aligned}
\]

where, \( \frac{1}{\rho^2} = \frac{1}{\rho_1^2} + \frac{1}{\rho_2^2} \), \( 0 < \rho < 1 \).

Thus, Eq.(53) is rewritten as follows:

\[
\begin{aligned}
\dot{L}_\varphi &\leq \frac{1}{2\beta} e^T \left( -Q + \frac{1}{\rho^2} PBB^T P \right) e + \hat{\beta}_\rho e^T PBR^{-1} B^T Pe + \hat{\beta}_\rho e^T PBR^{-1} B^T Pe + \\
&- \hat{\beta}_\rho e^T PBR^{-1} B^T Pe - \frac{1}{\beta_{\min 1}} (\Delta \varphi + \Delta \delta) \text{sign}(\hat{g}(x)) B^T Pe + \frac{\Delta \varphi + \Delta \delta}{\beta_{\min 1}} \text{sign}(\hat{g}(x)) B^T Pe \\
&+ \frac{1}{\beta_{\min 2}} e^T PB (D - \hat{D}) + \frac{1}{\beta_{\min 2}} (\nu - \hat{\nu}) B^T Pe e^T PB + \frac{1}{\gamma_{\rho}} \hat{\beta}_\rho \beta_{\rho} + \frac{1}{\gamma_{D}} \hat{D}_\rho \hat{D}_\rho + \frac{1}{\gamma_{\nu}} \hat{\nu} \hat{\nu}
\end{aligned}
\]

(54)

Using the adaptation laws (44)-(46), Eq.(54) is rewritten as follows:

\[
\dot{L}_\varphi \leq -\frac{1}{2\beta} e^T Q e + \frac{1}{2\beta \rho^2} e^T PBB^T Pe = -\frac{1}{2} \beta_\rho e^T Pe + \frac{1}{2\rho^2} \beta_\rho e^T PBB^T Pe
\]

\[
\Leftrightarrow \dot{L}_\varphi \leq -\frac{1}{2} \left| \beta_\rho \right| e^T Q e + \frac{1}{2} \left| \beta_\rho \right| \omega^2
\]

where, \( \frac{1}{\rho^2} e^T PBB^T Pe \leq \omega^2 \) with \( \omega^2 \) is the upper bound value of \( \frac{1}{\rho^2} e^T PBB^T Pe \). The above equation cannot be used to conclude the system stability. Hence, it needs to be integrated from \( t = 0 \) to \( t = T \):

\[
L_\varphi \left( 0 \right) - L_\varphi \left( T \right) + \frac{1}{2} \left| \beta_\rho \right| \int_0^T \omega^2 dt \geq \frac{1}{2} \left| \beta_\rho \right| \int_0^T e^T Q e dt
\]

(55)

where, \( L_\varphi \left( 0 \right) = \frac{1}{2\beta} e^T \left( 0 \right) Pe \left( 0 \right) + \frac{1}{2\beta \gamma_{f}} \Phi_x^T \left( 0 \right) \Phi_x \left( 0 \right) + \frac{1}{2\gamma_{f}} \Phi_x^T \left( 0 \right) \Phi_x \left( 0 \right) + \frac{1}{2\gamma_{f}} \hat{\beta}_{\rho} \beta_{\rho} \left( 0 \right)+ \\
+ \frac{1}{2\gamma_{D}} \hat{D}^T \left( 0 \right) \hat{D} \left( 0 \right) + \frac{1}{\gamma_{\nu}} \hat{\nu}^T \left( 0 \right) \hat{\nu} \left( 0 \right). \) The value \( L_\varphi \left( T \right) \) is always positive, and thus Eq. (55) is determined as:
\[ L_\nu(0) + \frac{1}{2} \beta_\nu \int_0^T \omega^2 dt \geq \frac{1}{2} \beta_e \int_0^T e^r \mathbf{Q} e dt \geq 0 \]  \hspace{1cm} (56)

The proof is now completed. Figure 3 illustrates the block diagram of the proposed control 2. The control process is similar to the Figure 2. However, the dead-zone band-delayed model and a new modified Riccati-like equation are added for calculation.

**Remark 2.** As a similar way as Remark 1, the proposed adaptive law in (45) helps to reduce the energy consumption in control inputs.

**Remark 3.** In [18], the control process using approach used fuzzy model to handle the dead-zone. However, this approach requires higher computational burden consumed more time for calculation because of the overlap of the iteration parameters. The Nussbaum-type function was used to support this process. However, this application affects the system stability control and the delay time delay issues was not considered solved thoroughly.

**Remark 4.** In [19], the model of neural networks was applied for the dead-zone problem. However, the dead-zone model in [19] is was a simple form, which following the classical model. In addition, the delayed time delay issue was not solved in [19].

**Remark 5.** In [20], the dead-zone model was similar to the model in [19]. The delayed time delay of actuator was not mentioned in [19]. Hence, the proposed controllers in [19,20] were not obtained the required precision in control level for a complicated complex system. The delay and dead-zone of oil in hydraulic system [20] was impacted by pressure in valve and the pipe line.

**Remark 6.** In [21], the dead-zone model was still formed like the classical model. The inverse model was used with the adaptation laws for control. This inverse model was the main disadvantage of the control system [21], which increased the time of calculation time and the convergence of the desired system error. The delayed time was not taken into consideration added in [21].

**Remark 7.** In [22], the classical model of dead-zone was still applied. The prescribed performance was used for control defining the convergence of the error. However, the obtained results were similar like the results in [21] with the properties as mentioned discussed above.

**Remark 8.** In [23], the dead-zone model was developed based on the classical model. The time delay issue was also mentioned discussed in the model [23]. However, this approach preserves the disadvantages as that of disadvantages of the proposed model was remain as shown in the classical model. This will be pointed out in the next section, and the control [23] is will be also compared control with the proposed control in this study.

### 3. Simulation and Discussion

#### 3.1. Model of Simulation
In this work, the vehicle seat suspension system with MR damper studied in [13] is adopted to demonstrate the effectiveness of the proposed controller. The mechanical model of the system is shown in Figure 4. The governing equations of the system are derived as follows:

\[
m_s \ddot{x}_s = -k_s (x_s - x_0) - c_s (\dot{x}_s - \dot{x}_0) + k_1 (x_1 - x_s) + c_1 (\dot{x}_1 - \dot{x}_s) + F_{MR} \tag{57}
\]

\[
m_1 \ddot{x}_1 = -k_1 (x_1 - x_s) + c_1 (\dot{x}_1 - \dot{x}_s) \tag{58}
\]

The above equations can be rewritten using the state variables as follows:

\[
\begin{align*}
\dot{x}_{11} &= \dot{x}_x = x_{22} \\
\dot{x}_{22} &= f_{11}(x_{11}, x_{22}, x_{33}, x_{44}) + g_{11}(x_{11}, x_{22}, x_{33}, x_{44}) \mu \\
\dot{x}_{33} &= \dot{x}_1 = x_{44} \\
\dot{x}_{44} &= f_{22}(x_{11}, x_{22}, x_{33}, x_{44})
\end{align*}
\tag{59}
\]

where,

\[
\begin{align*}
f_{11}(x_{11}, x_{22}, x_{33}, x_{44}) &= -\frac{k_s}{m_s} (x_{11} - x_0) - \frac{c_s}{m_s} (x_{22} - \dot{x}_0) + \frac{k_1}{m_s} (x_{33} - x_{11}) + \frac{c_1}{m_s} (x_{44} - x_{22}), \\
g_{11}(x_{11}, x_{22}, x_{33}, x_{44}) &= \frac{1}{m_s}, \quad u = F_{MR}, \quad f_{22}(x_{11}, x_{22}, x_{33}, x_{44}) = -\frac{k_1}{m_1} (x_{33} - x_{11}) - \frac{c_1}{m_1} (x_{44} - x_{22})
\end{align*}
\]

\[w = \begin{bmatrix} w_1 \\ w_2 \\ w_3 \\ w_4 \end{bmatrix} = \begin{bmatrix} f(x) \\ g(x) \end{bmatrix}
\]

\[u = \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}
\]

\[\Phi_f = -\frac{\gamma_f}{\beta} B^T P e_u \quad \Phi_g = -\frac{\gamma_g}{\beta} B^T P e_u\]

\[\dot{P} = -\frac{\gamma}{\beta} P B^T P e_u - e_u P B \begin{bmatrix} f(x) \\ g(x) \end{bmatrix} - e_u P B \begin{bmatrix} f(x) \\ g(x) \end{bmatrix}
\]

\[\begin{align*}
\dot{x}_x &= f_{11}(x_{11}, x_{22}, x_{33}, x_{44}) + g_{11}(x_{11}, x_{22}, x_{33}, x_{44}) \mu \\
\dot{x}_1 &= f_{22}(x_{11}, x_{22}, x_{33}, x_{44})
\end{align*}
\]

Figure 3. Block diagram of the proposed algorithm 2.
It is notice that $x_{11}, x_{22}, x_{33}, x_{44}$ are variables related to $x_s$ and $x_1$ of the system (57)-(58).

**Figure 4.** Mechanical model of the seat suspension.

From Figure 1, the dead-zone for MR damper or generally MR device can be represented in Figure 5(a,b,c). It is remarked that there is also another phenomenon namely hysteresis in MR device. Normally, hysteresis phenomenon exists in the model of MR device. This phenomenon is explained as a nondegenerate input-output closed curve, which defines the frequency of excitation towards a DC signals [37], or a discontinuous jump in the force-velocity response [31]. The hysteresis phenomenon leads to some problems such as high tracking error, unwanted harmonics, and instability of the system. The difference between the dead-zone and hysteresis is that the value of the output will be zero for dead-zone, but a mean value for hysteresis. The disadvantages of the dead-zone phenomenon are quite similar to the hysteresis. Hence, the proposed controller for dead-zone problem is also true for the hysteresis issue.

![Diagram](image-url)
Figure 5. Parameters of damping force for dead-zone model: (a) Fixed-frame model, (b) Real-frame model

From Eq.(5), the main control function shown in Figure 5(a) can be written as follows:

\[
\phi(f_{MR}) = \beta f_{MR} + \varphi(f_{MR}) + \delta(\dot{f}_{MR})
\]  

(60)

where, \( f_{MR} \) is the theoretical damping force of the damper. Using similar definitions used in [32], the parameters of Eq.(60) are defined as:

\[
f_{MR} = \frac{\chi^2}{\Omega} v, \quad \dot{f}_{MR} = \frac{\chi^2}{\Omega} \dot{v}
\]  

(61)

where, \( \chi \) (V.s/m) is the actuation constant related to the viscous coefficient, \( \Omega \) is the resistance of the circuit of the damper (\( \Omega \)), \( v \) (m/s) is the velocity of the damper. The function \( \varphi(f_{MR}) \) is defined as follows:

\[
\varphi(f_{MR}) = \begin{cases} 
-\frac{\chi}{\Omega} e_d & \text{if } v > \frac{e_d}{\chi} \\
0 & \text{if } -\frac{e_d}{\chi} < v < \frac{e_d}{\chi} \\
\frac{\chi}{\Omega} e_d & \text{if } v < -\frac{e_d}{\chi}
\end{cases}
\]  

(62)

where, \( e_d \) is the voltage applied to the MR damper. The value of \( \varphi(f_{MR}) \) can be understood as \( \Delta \varphi \). The value \( \delta(\dot{f}_{MR}) \) is determined based on the upper bound \( W \), which is defined as an offset of the damping force. This value can be understood as \( \Delta \delta \). For the MR damper, the value \( W \) is chosen considering the range of the minimum and maximum damping force shown in Figure 5(b). From Eqs. (60) and (61), the relationship between the damping force and main control input \( u \) can be expressed as:

\[
F_{MR} = \tilde{\beta}_\rho u + \varphi(f_{MR}) + 2W = \tilde{\beta}_\rho u + \Delta \varphi + \Delta \delta
\]  

(63)

From Eq. (63), the main control input \( u \) is obtained as follows:

\[
u = \frac{F_{MR} - \Delta \varphi - \Delta \delta}{\tilde{\beta}_\rho}
\]  

(64)

It is notice that in the practical application of the proposed controller, from the resistance of the damper, the voltage \( \delta \) will be changed to the current to generate the magnetic field which is
required to achieve the desired damping force of the MR damper. It is remarked that the controllers proposed in [23, 36] will be used as a comparative model to verify the effectiveness of the proposed controller in this study.

3.2. Simulation Results

In this simulation, the performance of the proposed control declared in theorems 1 and 2 will be demonstrated. To verify the effectiveness of the proposed control, it is compared with other state-of-the-art controllers proposed in [23] (compared control 1) and [36] (compared control 2). Computer simulation for the above MR damper is carried out to evaluate the performance of the controllers. The principal parameters of the MR damper, seat suspension system and the converting values from voltage to current are given in [13]. The parameters of the dead-zone model are firstly proposed, and then updated according to the learning progress of the fuzzy neural networks model. Two different excitations, i.e., the random bump road and random step wave road shown in Figure 6, are imposed to the seat suspension system. The maximum force of MR damper is designed to have 1000 N at 2 A. The fuzzy neural networks model is adapting online with the centroid vector given in [13]. The selected parameters of the proposed controller, the controller proposed in [23] and the controller proposed in [36] are listed in Table 1, Table 2, and Table 3, respectively. In this simulation, the initial states for the dynamic states are assigned as $[0.035 \ 2.5]$ and $[0.035 \ 2.5]$ for random bump road, and random step wave bump, respectively. The initial states for the observer are assigned as $[0.035 \ 0]$ for two excitations.
Figure 6. Excitation signals: (a) random bump road, (b) random step wave road

Table 1. Parameters of simulation of proposed control

<table>
<thead>
<tr>
<th>Description</th>
<th>Symbol</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Delayed time</td>
<td>( \tau )</td>
<td>0.1</td>
</tr>
<tr>
<td>Control function</td>
<td>( \delta(\dot{u}) )</td>
<td>0.07</td>
</tr>
<tr>
<td>Dead-zone performance</td>
<td>( \beta^+, \beta^- )</td>
<td>0.08; -0.08</td>
</tr>
<tr>
<td>Dead-zone performance</td>
<td>( \beta )</td>
<td>0.06</td>
</tr>
<tr>
<td>Dead-zone performance</td>
<td>( \beta_{\text{min}} )</td>
<td>0.65</td>
</tr>
<tr>
<td>Dead-zone performance</td>
<td>( \beta_{\text{min}1}, \beta_{\text{min}2} )</td>
<td>0.65</td>
</tr>
<tr>
<td>Riccati-like constant</td>
<td>( \rho )</td>
<td>0.65</td>
</tr>
<tr>
<td>Gain matrix</td>
<td>( K_i = [K_{i1}, K_{i2}]^T )</td>
<td>( K_i = [0.0001, 1]^T )</td>
</tr>
<tr>
<td>Constants</td>
<td>( \gamma_f, \gamma_g, \gamma_D, \gamma_\beta, \gamma_v )</td>
<td>700</td>
</tr>
<tr>
<td>Riccati-like matrix</td>
<td>( Q )</td>
<td>([-2; 0; -2])</td>
</tr>
<tr>
<td>Riccati-like matrix</td>
<td>( R )</td>
<td>0.1</td>
</tr>
</tbody>
</table>

Table 2. Parameters of simulation of compared control 1 [23]

<table>
<thead>
<tr>
<th>Description</th>
<th>Symbol</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Delayed time</td>
<td>( \tau )</td>
<td>0.1</td>
</tr>
<tr>
<td>Gain matrix</td>
<td>( K_i = [K_{i1}, K_{i2}]^T )</td>
<td>( K_i = [0.3, 1]^T )</td>
</tr>
<tr>
<td>Control function</td>
<td>( \delta(\dot{u}) )</td>
<td>0.07</td>
</tr>
<tr>
<td>Dead-zone performance</td>
<td>( \beta )</td>
<td>0.06</td>
</tr>
<tr>
<td>Dead-zone performance</td>
<td>( \beta_{\text{min}} )</td>
<td>0.65</td>
</tr>
<tr>
<td>Dead-zone performance</td>
<td>( \beta^+, \beta^- )</td>
<td>0.08; -0.08</td>
</tr>
<tr>
<td>Riccati-like constant</td>
<td>( \rho )</td>
<td>0.65</td>
</tr>
<tr>
<td>Riccati-like matrix</td>
<td>( Q )</td>
<td>([-2; 0; -2])</td>
</tr>
<tr>
<td>Riccati-like matrix</td>
<td>( R )</td>
<td>0.1</td>
</tr>
</tbody>
</table>

Table 3. Parameters of simulation of compared control 2 [36]

<table>
<thead>
<tr>
<th>Description</th>
<th>Symbol</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Dead-zone performance</td>
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<td>1</td>
</tr>
<tr>
<td>Dead-zone performance</td>
<td>( \Omega )</td>
<td>1.7</td>
</tr>
<tr>
<td>Dead-zone performance</td>
<td>( \bar{\alpha} )</td>
<td>1000</td>
</tr>
</tbody>
</table>

Simulation results will include the displacement of seat and force control input. These dynamic responses have important roles in evaluating the performance of the comparative controllers in terms of vibration suppression and energy consumption. The responses of the system under the inputs of the first proposed control (\textit{theorem 1}) and the second proposed control (\textit{theorem 2}) are shown in Figures 7 and 8 and Figures (9) and 10, respectively. The displacements of the seat
under the controllers are shown in Figures 7 and 9. In Figure 7, the displacement of the seat under the inputs of the three controllers are fairly good. It is shown from Figure 7(a1) that the magnitude of vibration is significantly decreased after applying three controllers. The variations of control inputs and observers are shown in Figure 7(a2). This figure shows that the performances of the three controllers satisfy the observer’s requirements. However, the performance of the proposed controller is better than that of the comparative controllers, as shown in Figure 7(a3) and Figure 7(a4). For the excitation of random pump road, the results are shown in Figures 7(b1), 7(b2), 7(b3), and 7(b4). It is clearly seen from Figure 7(b3) and 7(b4) that the proposed controller provides better performance than the comparative controllers. The control input $u$ is shown in Figure 8. It is clearly observed that the control energy of the proposed controller is less than that of the comparative controllers. This results directly prove that the proposed controller requires less power consumption, but provides better control performance. This merit is obtained based on the use of the adaptation law for the dead-zone problem, as discussed in remark 1. The flexible adaptation values of the update law $\hat{\beta}$ in (25) and other adaptation laws of the proposed controller 1 and the other laws improve the system performance significantly with application of the proposed controller 1. This is also demonstrated that the affection effects of the dead-zone problem is limited can be suppressed.

For the excitation of random step wave road, the results are shown in Figures 9(a1)-9(a4). Similar results can be achieved for the random bump road, which verify better control performance of the proposed controller compared to other controllers. In addition, the vibration amplitude obtained by the proposed controller is less than that of the valued obtained by the comparative controllers, as shown in Figures 9(a3,a4,b3,b4). This results directly verify that the ride comfort of the driver can be enhanced by implementing the proposed controller. Figure 10 presents control inputs of the three controllers for two different road conditions. It can be seen that the input powers of the proposed controller less than the comparative controllers. Thus, the proposed controller provides better vibration suppression performance and less power consumption than the comparative controllers. The updated values of the update law $\hat{\beta}$ in (45), $\nu$ in (46), and the other adaptation laws of the proposed controller 2 improve the system performance under both
dead-zone and delayed time problem with application of the proposed controller 2. This is also show that the affection demonstrated that the effects of both dead-zone problem and delayed time and time delay have been suppressed is limited.

From the above results, the compared control 1 is not flexible in calculation, as shown discussed in remarks (1,2,8). It is also remained the main properties of the classical model of dead-zone. In the compared control 2, the simple calculation based on the classical dead-zone model is does still not solve the disadvantages as well. The compared control 2 is designed to only seek the tuning the parameters following the time scale only. In addition, the sine function in the derived functions of tuning method is one of the obstructed calculations. This is considered based on from the property of the system which does is not always follow a fixed model of dead-zone function. Then, the time convergence of the compared control 2 is also longer than the other controllers as shown in [36]. The simulation results verified are show that the proposed controller has overcome the disadvantages that the compared controls (1,2) encountered; and decreased the effects affection of the dead-zone and time delay in minimally to minimum.
Figure 7. Simulation results of the displacement at the seat ($x_s$): (a1) (a2)(a3)(a4) random bump road, (b1) (b2)(b3)(b4) random step wave road.
Figure 8. Simulation results of force control $u$: (a1)(a2) random bump road, (b1)(b2) random step wave road
Figure 9. Simulation results of the displacement at the seat \( x_s \): (a1) (a2) (a3)(a4)-random bump road; (b1) (b2) (b3) (b4)-random step wave road
Figure 10. Simulation results of force control $u$: (a1)(a2) random bump road, (b1)(b2) random step wave road

4. Conclusion

In this study, a new adaptive controller was proposed based on a type 2 fuzzy model integrated with a neural network to resolve the dead-zone band and time delay issues of actuators. The proposed approach preserves many merits of many traditional controllers and adaptive techniques, including type 2 fuzzy model, adaptive control laws based on the sliding mode controller, $H$-infinity control technique and modified Riccati-like equation. The new model of dead-zone band has overcome the disadvantage of the classical model of control, especially applying for smart materials such as piezoelectric, magneto-rheological fluid. The proposed dead-zone model is also can be applied for actuators made by using materials possessing high time delay delayed time parameters. Two theorems have been presented: one for dead-zone problem, and the other for both dead-zone problem and time delayed time. The combination between the values of the dead-zone band and the Riccati-like equation has improved the robustness properties of the controllers.

The stability and convergence of the proposed controllers are proved rigorously based on Lyapunov stability criterion. The effectiveness of the proposed controllers is validated through the computer simulation for a vehicle seat suspension system and comparing with two other state-of-the-art controllers. From the simulation results, it was demonstrated that the proposed controller provides better control performance with less power consumption compared to the two counterpart controllers. This performance enhancement indicates the better ride comfort of the vehicle seat. It should be mentioned that the proposed controller can be straightforwardly
applicable for many control systems activated by smart material actuators such as shape memory alloys, piezoelectric materials, magnetostrictive materials and magnetorheological elastomers. The future works of this study will concentrate on extending the applications of the proposed control approaches for real systems featuring smart materials such as seat suspension, energy harvesting system, mount vibration control.

Acknowledgement

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Declaration of conflicting interest

The authors declare that there is no conflict of interest.

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Nomenclature

\[ H^j_i (i = 1, ..., n; j = 1, ..., m) \] Fuzzy sets

\[ m \] Number of rules

\[ a^i_j \] Interval sets

\[ g_f \] Defuzzied output

\[ \Phi^f_i, \Phi^f_r \] Weighted firing strength vectors

\[ f(x) \in R^n, \ g(x) \in R^n \] Two unknown non-linear function vectors

\[ \phi_u (t) = \beta u + \varphi(u) + \delta(\dot{u}), \ \phi_\delta (t) \in R^1 \] Control function

\[ \beta \] Parameter related slope of the dead-zone performance (a chosen constant gain)

\[ u \] Control input

\[ \varphi(u) \] Dead-zone function

\[ \Delta \varphi \] Maximum boundary of dead-zone model

\[ \delta(\dot{u}) \] Parameter related varied boundary of dead-zone model

\[ \Delta \delta \] Maximum value of varied boundary

\[ d(t) \in R^n \] An external disturbance vector

\[ \delta d \in R^n \] Upper bound of \( d(t) \)

\[ x = [x_1, x_2, ..., x_n]^T = [x_1, \dot{x}_1, ..., x_1^{(n-1)}]^T \in R^n \] The state vector of the system

\[ x_\tau = [x_1^{(t-\tau_1)} \ x_2^{(t-\tau_2)} \ ... \ x_n^{(t-\tau_n)}]^T \] The state with time delay
\( \Lambda_t, \Lambda_m(t) = \begin{bmatrix} z_m(t) & 0 \end{bmatrix}^T, \quad |z_m(t)| \leq \delta z_m \)  

The nonlinear function related time-varying of the system

\( \delta z_m \)  

Unknown positive constant

\[ x_d = [x_{d1}, x_{d2}, \ldots x_{dn}]^T \quad [x_{d1}, x_{d2}, \ldots x_{d(n-1)}]^T ]^T \in R^n \]  

Desired value

\[ e = x - x_d \]  

Error vector

\[ K = diag \{ K_1, \ldots, K_n \} \]  

Feedback gain matrix

\[ K_i = [K_{i1} \quad K_{i2}]^T \]  

Chosen vector such that all roots of the characteristic polynomial of

\[ A_i = A - BK_i^T, \quad i = 1, \ldots, n \] are in the open left half plane

\[ \hat{f}(x) = \Xi_f(x)\Phi_f \quad \hat{g}(x) = \Xi_g(x)\Phi_g \]  

Fuzzified functions of \( f(x) \) and \( g(x) \), respectively

\( \Delta f(x), \Delta g(x) \)  

The minimum approximation errors

\[ D = \Delta f + \Delta g(\beta u + \phi(u) + \delta(\dot{u})) \]  

The minimum approximation error related dead-zone phenomenon

\( u_1 \)  

The equivalent control

\( u_2 \)  

The first robustness control related the Riccati-like equation

\( u_3 \)  

The second robustness control related the dead-zone band, the Riccati-like equation and the minimum approximation error \( D \) of the system

\( u_4 \)  

The third robustness control related the delayed model

\( \dot{v} \)  

Parameter related delayed time

\( R \)  

The robust H infinity technique gain
\( \beta_{\text{min1}} \) Parameter related dead-zone model

\( \beta_{\text{min2}} \) Parameter related delayed model

\( \gamma_f, \gamma_g, \gamma_D, \gamma_D, \gamma_D \) Chosen constants

\( m_s \) Mass of seat

\( m_1 \) Mass of driver

\( k_s \) Stiffness coefficient of seat

\( k_1 \) Stiffness coefficient of torso

\( c_s \) Damping coefficient of seat

\( c_1 \) Damping coefficient of torso

\( x_{11}, x_{22}, x_{33}, x_{44} \) Variables related \( x_i \) and \( x_j \) of the system

\( f_{\text{MR}} \) Theoretical damping force of the damper

\( \chi \) The actuation constant related viscous coefficient (V.s/m)

\( \Re \) The resistance of the circuit of damper (\( \Omega \))

\( v \) The velocity of damper (m/s)

\( e_d \) The voltage applied to the damper

\( \Gamma \) The upper boundary \( \Gamma \) which is defined as an offset of the damping force, and this value can be seen as \( \Delta \delta \)

\( \theta \) The voltage to be applied to MR damper